

# Engineering Notes

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## Calculating Collision Probability for Arbitrary Space-Vehicle Shapes via Numerical Quadrature

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### I. Introduction

**M**ETHODS to compute space-vehicle collision probability have been developed and presented in earlier publications.<sup>1–5</sup> The contour integration method was developed to improve computational efficiency and handle arbitrary space-vehicle shapes.<sup>3</sup> Although the method has been applied successfully to related problems,<sup>4,5</sup> some analysts might be reluctant to use the method because it involves contour integration rather than the more familiar numerical quadrature. The current work improves on the previous method by reducing the contour integral to a one-dimensional integral and simplifying space-vehicle shape definition. This improved method was found to be easy to implement, computationally efficient, and accurate.

In computing collision probability, each space object is propagated to a point of closest approach. Assuming the relative velocity between the orbital objects at closest approach is very large compared to the effects of their relative acceleration, the relative velocity can be considered constant during the encounter period. A special method was developed to treat cases involving nonlinear relative motion; however, it will not be discussed here.<sup>6,7</sup>

The collision probability is the integral of the relative position error probability density over the volume swept out by the combined hardbody of the two space objects. After integrating the relative position error along the relative velocity vector direction, one obtains a two-dimensional integral over the projection of the combined hardbody on to a plane normal to the relative velocity vector. The previous work reduced this two-dimensional integral to a one-dimensional contour integral about the perimeter of the projected hardbody area.<sup>3</sup> This was achieved by performing a scale change that symmetrized the position error probability density. Because the contour integral is one dimensional, it requires fewer numerical integration steps than the two-dimensional method for comparable accuracy. It was shown that this simplified form significantly reduces computation time<sup>3,4</sup> and enables treatment of asymmetric hardbody shapes including tethers.<sup>5</sup>

The current implementation involves shifting the origin of the coordinate system to transform the contour integral to a one-dimensional angular integral. Because the integrand contains the definition of the perimeter in functional form, the same integral can be used for any desired hardbody shape. One needs only to define the desired hardbody shape mathematically. This makes implementation easier because definition of the hardbody shape is performed separately and does not alter other portions of the processing.

The algorithm was implemented in a computer program and tested with known results to validate accuracy. It is particularly useful in situations involving numerous evaluations of collision probability, such as maneuver optimization.<sup>4</sup>

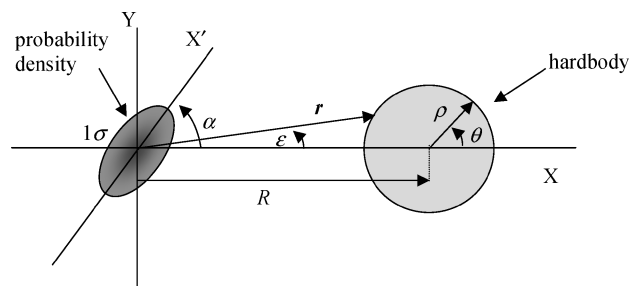
### II. Analysis

Once two objects are identified to have collision risk, they are propagated to a time near the closest approach time, such that their relative motion is linear. The relative position and velocity near closest approach are used to determine the closest approach distance and the encounter frame. The positive  $z$  axis of the encounter frame is opposite in direction to the relative velocity vector. The positive  $x$  axis of the encounter is directed from the first object to the second object. The  $y$  axis completes the right-handed system.

The combined hardbody radius is obtained by adding the respective hardbody radii of the two space objects. Figure 1 illustrates the position error covariance probability density, combined hardbody, and positions of the respective space objects in the encounter plane. The probability density is centered on the first space object, which is located at the origin of the coordinate system. The axis having the largest position error standard deviation  $\sigma_x$  axis makes an angle  $\alpha$  with the  $x$  axis, as illustrated in Fig. 1. The combined hardbody is centered on the second space object located on the  $x$  axis. The collision probability is the integral of the probability density over the hardbody area as presented in Eq. (1), where  $P$  is the collision probability,  $r$  is the distance to the hardbody perimeter, and  $\sigma$  is the symmetrized position error standard deviation. Details of the derivation of the probability integral can be found in the original paper<sup>3</sup>:

$$P = \frac{1}{2\pi} \oint_{\text{perimeter}} \left[ 1 - \exp\left(\frac{-r^2}{2\sigma^2}\right) \right] d\epsilon \quad (1)$$

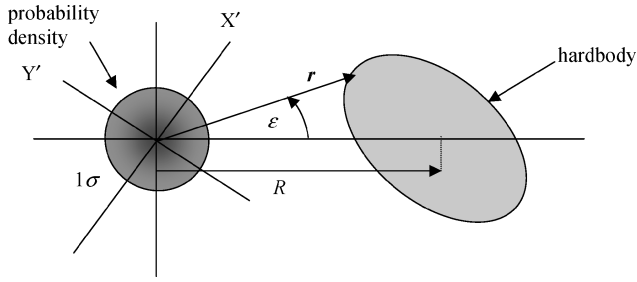
Equation (1) can be converted to a conventional definite integral by changing to polar coordinates centered in the hardbody volume. Let the new coordinate system be centered at  $(R, 0)$ . Points along the



**Fig. 1 Hardbody and combined position probability density in encounter plane before symmetrization.**

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**Fig. 2 Deformed hardbody and probability density after symmetrization.**

hardbody perimeter are defined as

$$x = R + \rho \cos(\theta), \quad y = \rho \sin(\theta) \quad (2)$$

where  $\rho$  is the radial position of a perimeter point and  $\theta$  is the angular position as defined in Fig. 1.

A coordinate rotation and scale change transform perimeter points to the symmetrized coordinate frame, as illustrated in Fig. 2:

$$\begin{pmatrix} x' \\ y' \end{pmatrix} = \begin{pmatrix} 1 & 0 \\ 0 & f \end{pmatrix} \begin{bmatrix} \cos(\alpha) & \sin(\alpha) \\ -\sin(\alpha) & \cos(\alpha) \end{bmatrix} \begin{pmatrix} x \\ y \end{pmatrix} \quad (3)$$

where  $f$  is the ratio of the  $x$  axis  $\sigma_{x'}$  to  $y$  axis  $\sigma_{y'}$ , in the diagonal frame.

The contour integration parameter in Eq. (1)  $\varepsilon$  is related to  $\theta$  by the relation

$$\tan(\varepsilon) = \frac{y'}{x'} = \frac{f\rho \cos(\alpha) \sin(\theta) - Rf \sin(\alpha) - f\rho \sin(\alpha) \cos(\theta)}{R \cos(\alpha) + \rho \cos(\alpha) \cos(\theta) + \rho \sin(\alpha) \sin(\theta)} \quad (4)$$

which can be written in compact form as

$$\tan(\varepsilon) = \frac{f\rho \sin(\theta - \alpha) - Rf \sin(\alpha)}{R \cos(\alpha) + \rho \cos(\theta - \alpha)} \quad (5)$$

The derivative of Eq. (5) with respect to  $\varepsilon$  yields

$$\frac{d\varepsilon}{d\theta} = \cos^2(\varepsilon) \frac{d[\tan(\varepsilon)]}{d\theta} = \frac{f\rho^2 + f\rho R \cos(\theta) + Rf\rho' \sin(\theta)}{r^2} \quad (6)$$

where

$$\begin{aligned} r^2 &= \sqrt{x'^2 + y'^2} = [R + \rho \cos(\theta)]^2 [\cos^2(\alpha) + f^2 \sin^2(\alpha)] \\ &\quad + \rho^2 \sin^2(\theta) [\sin^2(\alpha) + f^2 \cos^2(\alpha)] \\ &\quad + 2\rho(1 - f^2) \cos(\alpha) \sin(\alpha) \sin(\theta) [R + \rho \cos(\theta)] \end{aligned} \quad (7)$$

and

$$\rho' = \frac{d\rho}{d\theta}$$

Using Eq. (6) in Eq. (1) transforms Eq. (1) from a contour integral to a definite integral given by

$$\begin{aligned} P &= \frac{1}{2\pi} \int_0^{2\pi} \left[ \frac{f\rho^2 + Rf\rho \cos(\theta) + Rf\rho' \sin(\theta)}{r^2} \right] \\ &\quad \times \left[ 1 - \exp\left(\frac{-r^2}{2\sigma^2}\right) \right] d\theta \end{aligned} \quad (8)$$

If  $R = \rho$ ,  $r$  can equal zero, yet the integrand remains finite. In this case, the exponential term in the integrand can be expanded. Keeping the first few terms in the expansion, the integrand becomes

$$[f\rho^2 + Rf\rho \cos(\theta) + Rf\rho' \sin(\theta)](1/2\sigma^2 - r^2/8\sigma^4 \dots) \quad (9)$$

Equation (8) provides an accurate method for computing collision probability very efficiently. For spherical hardbodies,  $\rho$  is constant, and the  $\rho'$  term vanishes. If  $\rho$  varies as a function of  $\theta$ , the  $\rho'$  term is retained. Simple numerical methods enable the integration to be accomplished without the evaluation of trigonometric functions. Therefore, the evaluation of Eq. (8) is extremely efficient. No approximations were used in obtaining Eq. (8). The definite integral form of Eq. (8) can be used in numerical integration packages, thus enabling easy implementation.

Numerical precision can be improved by splitting the integrands in Eqs. (1) and (8):

$$P = \frac{1}{2\pi} \oint_{\text{perimeter}} d\varepsilon - \frac{1}{2\pi} \oint_{\text{perimeter}} \exp\left(\frac{-r^2}{2\sigma^2}\right) d\varepsilon \quad (10)$$

$$\begin{aligned} P &= \frac{1}{2\pi} \oint_{\text{perimeter}} d\varepsilon - \frac{1}{2\pi} \int_0^{2\pi} \left[ \frac{f\rho^2 + Rf\rho \cos(\theta) + Rf\rho' \sin(\theta)}{r^2} \right] \\ &\quad \times \exp\left(\frac{-r^2}{2\sigma^2}\right) d\theta \end{aligned} \quad (11)$$

The first term in Eqs. (10) and (11) is easily evaluated based on the location of the origin with respect to the hardbody area:

$$\frac{1}{2\pi} \oint_{\text{perimeter}} d\varepsilon = \begin{cases} 0, & R > \rho \\ 1, & R < \rho \end{cases} \quad (12)$$

Equation (9) could be used for the special case of  $R = \rho$ .

### III. Numerical Results

Equation (8) was implemented in a computer program and used to evaluate the collision probability for a set of 28 space object test cases having spherical hardbodies, as illustrated in Table 1. For these cases,  $\rho'$  is zero, because  $\rho$  is constant. The same cases were evaluated using contour integration in Eq. (1). Results from the two methods were in agreement to within  $3 \times 10^{-6}\%$  error using only 200 integration steps in Eq. (8). Table 1 also contains closest approach distance and combined hardbody radius.

These same cases were used to check the validity of the  $\rho'$  term in Eq. (8) by shifting the origin of  $\rho$  as illustrated in Fig. 3. In this

**Table 1 Comparison of results**

Closest approach, km	Hardbody radius, km	Collision probability	Difference between methods, %
10.30595	0.261429	$7.11 \times 10^{-05}$	$2.57 \times 10^{-06}$
11.7910878	0.200077	$4.32 \times 10^{-05}$	$2.57 \times 10^{-06}$
5.3875133	0.271668	$1.30 \times 10^{-04}$	$2.56 \times 10^{-06}$
7.4893021	0.200515	$5.52 \times 10^{-05}$	$2.57 \times 10^{-06}$
34.5054014	0.265202	$1.25 \times 10^{-04}$	$2.58 \times 10^{-06}$
3.1965724	0.368366	$3.05 \times 10^{-04}$	$2.55 \times 10^{-06}$
75.3755093	0.200401	$3.56 \times 10^{-06}$	$2.56 \times 10^{-06}$
59.824544	0.200479	$5.54 \times 10^{-06}$	$2.56 \times 10^{-06}$
67.4297582	0.223129	$4.55 \times 10^{-06}$	$2.55 \times 10^{-06}$
46.0314966	0.258091	$4.01 \times 10^{-06}$	$2.57 \times 10^{-06}$
86.6519575	0.367134	$1.06 \times 10^{-05}$	$2.57 \times 10^{-06}$
78.4784936	0.200167	$5.08 \times 10^{-06}$	$2.58 \times 10^{-06}$
23.68313	0.300531	$1.00 \times 10^{-05}$	$2.58 \times 10^{-06}$
48.9289256	0.200404	$3.14 \times 10^{-06}$	$2.56 \times 10^{-06}$
123.6147301	0.200809	$2.94 \times 10^{-06}$	$2.56 \times 10^{-06}$
80.1446518	0.200866	$6.00 \times 10^{-06}$	$2.56 \times 10^{-06}$
45.2565622	0.200011	$5.61 \times 10^{-06}$	$2.59 \times 10^{-06}$
66.0937136	0.200071	$1.74 \times 10^{-06}$	$2.57 \times 10^{-06}$
78.3123166	0.200753	$2.60 \times 10^{-06}$	$2.56 \times 10^{-06}$
29.4894323	0.200135	$4.79 \times 10^{-06}$	$2.57 \times 10^{-06}$
59.9174311	0.261378	$5.21 \times 10^{-06}$	$2.56 \times 10^{-06}$
42.6579135	0.203766	$6.51 \times 10^{-06}$	$2.59 \times 10^{-06}$
140.0742924	0.200206	$2.54 \times 10^{-06}$	$2.58 \times 10^{-06}$
45.2052324	0.200079	$3.27 \times 10^{-06}$	$2.56 \times 10^{-06}$
85.4181172	0.200388	$3.04 \times 10^{-06}$	$2.56 \times 10^{-06}$
33.9892239	0.200960	$2.57 \times 10^{-06}$	$2.57 \times 10^{-06}$
17.6995351	0.200603	$3.51 \times 10^{-06}$	$2.57 \times 10^{-06}$
3.9057833	0.299000	$1.32 \times 10^{-05}$	$2.55 \times 10^{-06}$

